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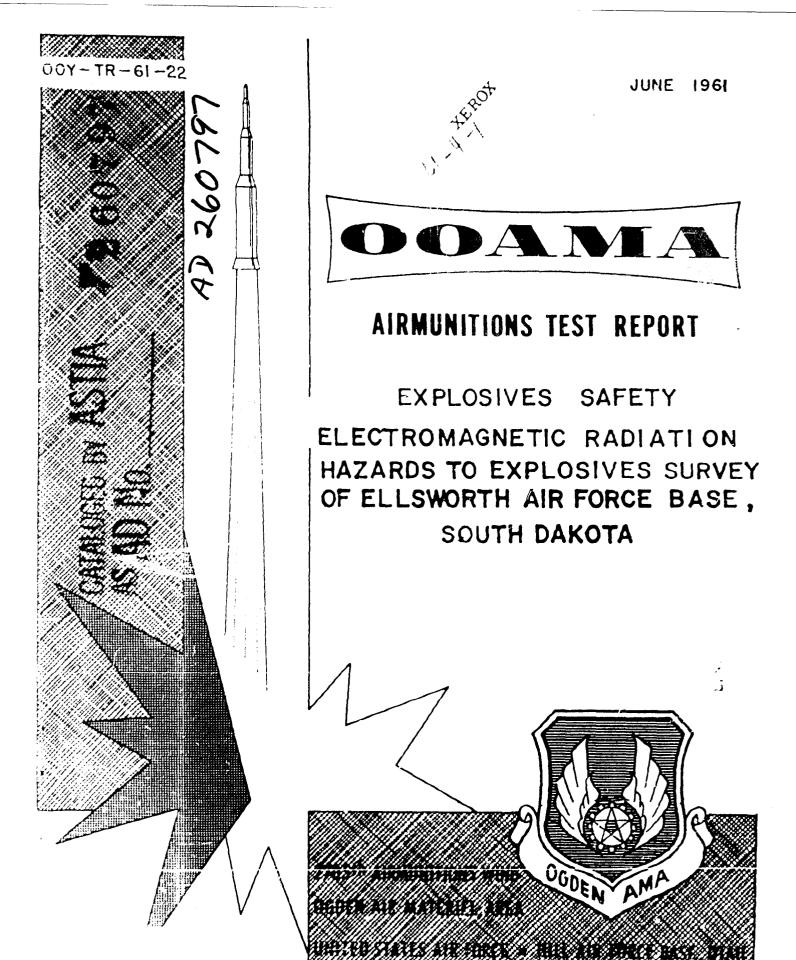
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ELECTROMAGNETIC RADIATION HAZARDS TO EXPLOSIVES

SURVEY OF ELLSWORTH AIR FORCE BASE, SOUTH DAKOTA

by

Harold R. Laughter

PUBLICATION REVIEW

This report has been reviewed and is approved

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JUNE 1961

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NOTICES

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The conclusions and recommendations made in this report are not to be considered directive in nature. This type information becomes official only when published in Technical Orders and/or other applicable Air Force publications.

ADMINISTRATIVE DATA

PURPOSE OF TEST:

The purpose of this survey was to determine the extent of electromagnetic radiation hazards to explosives storage, handling and shipping areas in the vicinity of the A.C. and W. site at Ellsworth Air Force Base. South Dakota.

ITEMS TESTED:

Ten S-68 Squibs, FSC 1375-035-6021-M846 Ten E-81 Squibs, FSC 1375-041-1312-M138

Five Rounds of 20 MM, M55 (TP) Ammunition, FSC 1305-529-7208-

A891 (Electric Primers)

Two 2.75 Inch FFAR Rockets, FSC 1340-038-8192-H500 Two MK 165 Igniters, FSC 1340-309-5095-H403

Two MK 15 Igniters

Two GAR 1 and 2A Motor and Nozzles

SECURITY CLASSIFICATION:

Unclassified

DATE TEST COMPLETED:

January 1961

SURVEY CONDUCTED BY &

OOAMA (OOYSS) - 2705th Airmunitions Wing

Project Engineer: Harold R. Laughter

ABSTRACT

The purpose of this survey was to determine the extent of electromagnetic radiation hazards to explosives in storage, handling and shipping areas in the vicinity of the A.C. and W. site at Ellsworth Air Force Base, South Dakota. Field strength measurements of the main beams from the AN/FPS-20 and AN/MPS-14 radars were made at various locations in the aforementioned areas and along the transportation route. Various electrically initiated explosive items were exposed to the main beam of the radars, following the highest terrain possible into the AN/MPS-14 radar system. It was concluded that a degree of hazard does exist along the transportation route to the munitions storage area.

TABLE OF CONTENTS

																									F	AGE
Notices	•		•		•	•	•	•	•	0	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	ii
Administrat																										
Abstract .																										
Table of Co																										
Introductio																										1
Description																										1
Figure 1																										
Figure 2																										5
Figure 3																										_
Figure 4																										
Equipment .																										13
Figure 5																										19
Figure 6																										21
																										23
Figure 7																										13
Test Proced																										
Figure 8																										25
Figure 9																										27
Figure 10																										29
Figure 11																										31
Figure 12																										33
Table 1 .																										39
Conclusions																										14
Figure 13																										35
Figure 14		, ,	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	37
Recommendat	io.	מו	8	۰	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	15
Calculation	ıs		•		•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•		•	•	•	•	15
Dictal butto																										41

INTRODUCTION

This survey was initiated to obtain data and information necessary to determine the extent of the electromagnetic radiation hazards to explosive storage, handling and shipping areas in the vicinity of the A.C. and W site at Ellsworth Air Force Base, South Dakota.

This survey was conducted under the provision of a General Test Plan for Electromagnetic Radiation Hazards at Ellsworth Air Force Base. The test plan was prepared by the Explosive Evaluation Branch (OOYEE) and coordinated with the Safety Requirements Branch (OOYSS). This survey was conducted by the Safety Requirements Branch (OOYSS), 2705th Airmunitions Wing with support from the 43d Munition Maintenance Squadron at Ellsworth Air Force Base, 740th A.C. and W. Squadron at Ellsworth Air Force Base, and Central GEEIA Region at Tinker Air Force Base.

Electrically initiated explosive items and initiators were selected on the basis of their relative sensitivity to electromagnetic energy.

DESCRIPTION

The following is a description of the electrical characteristics of the items tested:

- 1. S-68 Squib, FSC 1375-035-6021-M846 resistance 1.37 ± 0.50 ohms, Maximum no-fire current 0.30 ampere.
- 2. S-75 Squib, FSC 1375-529-9301-M856 resistance 5.0 $\stackrel{+}{\underline{}}$ 1.0 ohms, 100% fire current 0.58 ampere.
- 3. E-81 Blasting Cap, FSC 1375-041-1312-M138 resistance 1.57 \pm 0.50 ohms, Minimum fire current 0.50 \pm 0.05 ampere.
- 4. 20 MM, M55 (TP) Ammunition, FSC 1305, contains primer M52A3, resistance 1000 to 5,000,000 ohms, fire voltage 115.
- 5. 2.75 FFAR Rocket, FSC 1340-038-8192-H500, igniter use a MK 1 Mod 0 Squib 0.30 all fire, 0.20 no fire, reisstance 1.0 ± 0.3 ohms.
- 6. Igniter, Mk 165 for 15KS1000 JATO, FSC 1340-309-5095-H403 contains an electrically activated glow plug resistance 0.25 ohms, operating current 25 amperes.

a chialing Mineral

- 7. Igniter, M15 for 16NS1000 JATO, contains two M-2 Squibs, resistance 0.75-1.25 ohms, minimum fire current 0.545 ampere.
- 8. Motor and Nozzle, GAR 1 and 2, with igniter M50 and fuze T1403E3. M50 igniter contains M107A squib resistance 0.70 ohms, 0.25 ampere maximum no fire current. T1403E3 fuze contains two M4 dimple motors 5-9 ohms resistance 1.00 ampere. Current 100% fire, 0.1 ampere maximum no fire current. Also contains two M52 detonators, 55 volts, 100% fire resistance 1000-10,000 ohms.

The DuPont S-68 and S-75 squibs and E-81 blasting caps are similar in construction. They consist of a copper shell closed at one end and sealed at the other end by crimpings around a 3/8 inch long rubber plug. Two copper wires are molded into the rubber plug. Across the inside ends of the wires is connected a resistance wire called a bridge wire and around the bridge wire is a sensitive explosive mixture known as the ignition bead which ignites the charge.

In the blasting caps the ignition bead ignites a filler charge which ignites a primer charge, this sets off the base charge. The squib or blasting cap is initiated by passing a small current through the lead wires which in turn passes through the bridge wire. The bridge wire is heated by the current which causes ignition at a predetermined temperature (Figures 1 and 2).

20 MM Ammunition consists of a cartridge case, which is usually brass, a projectile or bullet, a quantity of propellant powder and an electric primer. The electric primer consists of a brass cup with a hole in the cupped end, into which is assembled a brass button separated from the cup by a vinylite insulator, followed by a consolidated charge of a conductive primer mixture, a shellacked foil paper disk; finally, a thin gilding-metal cup support is pressed into the body. The insulator is red in color. The charge weighs 2.75 grams (max.). The electrical path is from the face of the button exposed through the hole in the cup, through the button, and then through the conductive mixture to the cup. The primer is initiated by electrical energy. (See Figure 3.)

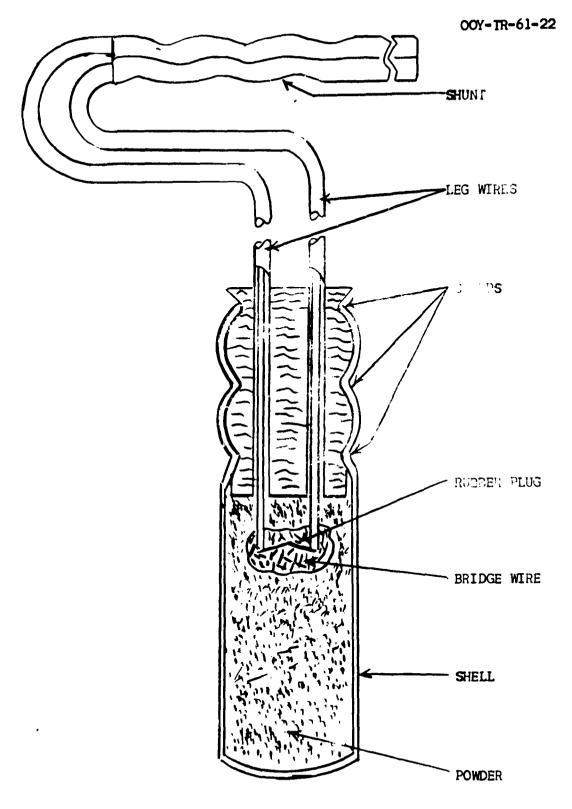


FIGURE 1. Cross-Section Drawing of Squib.

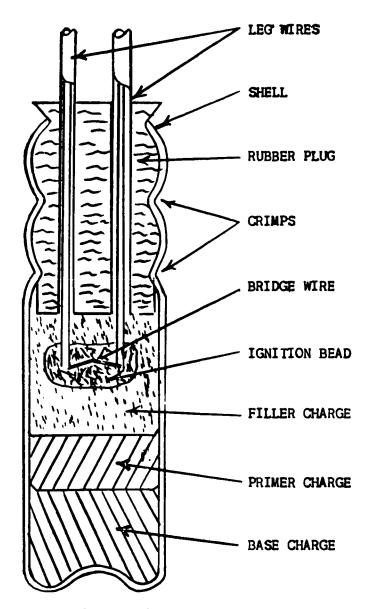


FIGURE 2. Cross-Section Drawing of Blasting Cap.

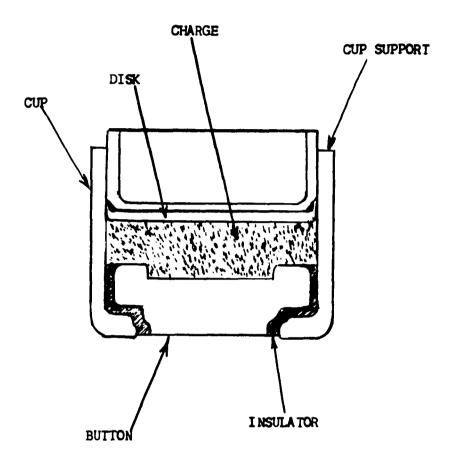


FIGURE 3. Electric Primer M52A381 - Enlarged.

The 2.75 Inch fin-stabilized aircraft rocket consists of a nose fuze, head, and motor. The motor consists essentially of an aluminum alloy tube containing propellant and an igniter, and has a nozzle-fin assembly attached to the aft end. The igniter is a metal case containing a mixture of black powder and magnesium powder, and an electric squib, and is located in the forward end of the motor. Two lead wires from the squib extend from the igniter passing through the perforation in the propellant grain to the nozzle plate where one lead wire is grounded to the nozzle plate. The other lead wire passes through the nozzle and is connected to the contact disk at the aft end of the rocket as the live contact.

The Mk 165 igniter consists of an electrical actuated glow plug that ignites black powder which in turn ignites the main charge contained in a steel mesh case.

The M15Al igniter consists of 55 grams of sustainer cast in the base of the igniter plug, and about 100 grams of pellets. Two M-2 squibs with aluminum sleeves are embedded in the sustainer propellant in the base of the igniter plug, with the firing end directed into the pellets. The pellets are contained in a one-quarter-inch wire mesh basket which has been dipped in a Tenite and rubber solution. Gotton packing is placed in the top of the wire basket to reduce pellet attrition during rough handling. The wire mesh basket is held onto the igniter plug by a strip key, and the bottom of the basket is sealed to the igniter plug with rubber cement. (See Figure 4.)

The GAR 1 and 2 motor case is made of steel and painted aluminum. The open end of the motor case is tapered and threaded to receive the adapter, which is threaded to receive the nozzle. These motors are approximately 36.74 inches long including nozzle, 5.848 inches diameter forward and 6.186 inches diameter at thrust flange. The M50 igniter used with the GAR 1 and 2 motor is 16.00 inches long and 0.62 inches in diameter. It contains two M107A squibs. The fuze for the GAR 1 and 2 contains two dimple motors M4, and two detonators, M52. The electrical connections to metal inclosure of the fuze consist of quarter-inch pins.

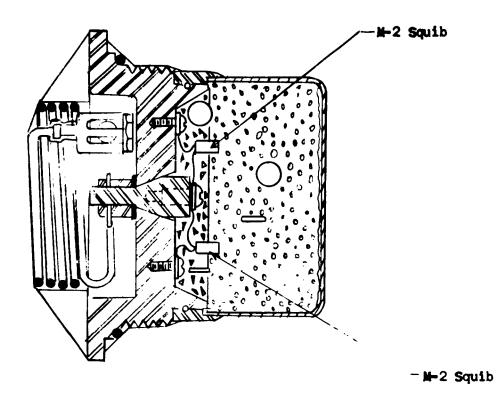


FIGURE 4. Igniter 16-NS-1000.

EQUIPMENT

The following equipment (Figure 5) was used during the survey:

- 1. Polorad field strength meter model FIM mounted in van (Figures 6 and 7).
- 2. Radar transmitter AN/MPS-14 (the AN/MPS-14 transmitter is identical to the AN/FPS-6 transmitter).
 - 3. Radar transmitter AN/FPS-20.
 - 4. Flat-bed 1-1/2 ton truck.
 - 5. Press type camera.

TEST PROCEDURES AND RESULTS

Field strength measurements were made in the munitions storage area at various locations among the igloos and warehouses. Attempts were made to take the readings in locations where the field strengths would be at a maximum and yet be in a position where explosive materials would be frequently handled. Readings were taken of transmitted signals from both the AN/FPS-20 and AN/MPS-14 radars. There was evidence of signal reinforcement. Reflections from the iron doors on the igloos and warehouses plus multipathing accounted for the reinforcement. In no event was the signal strength more than doubled by reflections and multipathing.

Additional field strength measurements were made along the transportation route to and from the storage area. The terrain features were such that the road dropped at approaches to the radars reducing the hazard. The original survey conducted, 23-27 January 1961, indicated relatively little hazard as the AN/MPS-14 radar antenna did not read below a minus 30 minutes. Subsequently the radar antenna was overhauled and an electrical alignment check showed the antenna nods to a minus 2 degrees as specified in the AN/MPS-14 specifications. Also, a field strength measurement was made at the Rushmore Air Force Station gate for information purposes.

The test samples, as listed in the administrative data and described previously, were arranged in various positions on a flat bed 1-1/2 ton truck. Arrangement of the samples was such as to present the more sensitive areas to the radars. Example, the GAR 1/2 motor was situated with the nozzle facing the radars; the nozzle functioning as a horn antenna. Squibs and blasting caps were taped on a sheet

of plywood with the lead wires forming various antenna configurations. Some of the antenna configurations are as follows: loop, rhombic, long wire and dipole. (See Figures 7..8. and 9.)

The truck with the test samples was backed into the concentrated radar beam following the highest possible terrain. (See Figures 10 and 11 for profile.) Exposures of the samples were made at 10 minute intervals in 200 feet increments from 4071.5 feet to 1271.5 feet, then in 100 feet increments to 271.5 feet from the center of the AN/MPS-14 radar tower. Field strength measurements were made at each interval.

A comparison of measured readings with the theoretical is shown in Figure 12 and Table 1. The measured values increased considerably over the theoretical as the instrument approached the radar. This is due to R.F. leakage, as the instrument was not shielded. Figure 12 also shows that the measured values do not fall into a smooth curve. These measurements are not made under free space conditions. The variations are accounted for and due to multipath and reflections. Subsequent measurements made, when checking the electrical alignment of the AN/MPS-14 radar antenna, was nearly double the theoretical values. This again is explained by multipathing and reflections.

None of the samples were initiated. This is attributed to two factors:

- 1. The AN/MPS-14 radar antenna was nodding to only a minus 30 minutes. The antenna has since been overhauled and nods to a minus 2 degrees.
- 2. The adverse weather conditions. During the survey temperatures were near zero and the wind velocity between 10 and 30 miles per hour.

CONCLUSIONS

Present operating procedures of the 43d Munitions Maintenance Squadron are safe. However, caution must be exercised when transporting electro-explosive devices to and from the munition storage area. All vehicles transporting electro-explosive devices or weapon systems containing electro-explosive devices must transport them in their original packaging and remain on Ramp Street to South Drive, then take South Drive directly to the Storage area. Terrain conditions reduce the hazard to some extent (Figure 13). However, there are areas which are potentially hazardous. These areas are marked on Figure 14.

RECOMMENDATIONS

- 1. It is recommended that all future surveys be conducted when the average ambient temperature is 70°F or above if it is possible.
- 2. It is recommended, that where Polarad field strength meter, Model FIM, is used, it be shielded.

CALCULATIONS

Original reading in db above 1 microvolt.

AN/MPS-14 radar transmitting on a frequency of 2.800 MC.

AN/FPS-20 radar transmitting on a frequency of 1300 MC.

Loss in cable 0 2800 MC = 3.5 db.

Loss in Cable @ 1300 MC - 2.7 db.

Free space conversion from polarad graphs @ 2800 MC = 22.1

Free space conversion from polarad graphs @ 1300 MC = 22.8

A. Received peak power measured from AN/MPS-14 radar in front of building number 9016 = 148 dbu

Total db u/m = 148 + 3.5 + 22.1 = 173.6 dbu/m

db = 20 LOG E

173.6 = 20 LOG E

LOGE = 8.68

E = 478.6 Volts/MeterPower Density = $\frac{E^2}{70}$

 Z^{O} = Impedance of Free Space = 120 M = 377

Power Density = $\frac{E^2}{70} = \frac{(478.6)^2}{377}$

= 0.0608 Watt/Qm² peak power

AN/MPS-14 Peak Power Output = 5,000,000 Watts Average Power Output = 3600 Watts

> Average Power Density = 0.0608 x $\frac{3600}{5.000,000}$ $= 0.0438 \times 10^{-3} \text{ Watts/cm}^2$

B. Theoretical Power Density.

Far Field Formula

$$W = \frac{P_T A_6}{4\pi d^2}$$

W = Power Density

Py = Transmitted Power

A6 - Power Gain of Antenna

d = Distance in Centimeters from Antenna

AN/MPS-14 RADAR

Power Output Average = 3600 Watts

Antenna Gain = 7400

Distance from AN/MPS-14 Radar Tower to Front of Bldg 9016 =

 $3775 \text{ feet} = 30.48 \times 3775 = 115062 \text{ CM}$

Power Density =
$$\frac{3600 \times 7400}{4\pi (115062)^2}$$

$$= 0.178 \times 10^{-3}$$

Near Field Corrections

$$F_f = F_1 \times F_2$$

1st Step

$$a = \frac{L_1}{\sqrt{2_4 \lambda}}$$

$$b = \frac{L_2}{\sqrt{2d \lambda}}$$

L₁ = Vertical dimension of antenna aperture in meters.

L₂ = Horizontal dimension of antenna aperture in meters.

d = Distance from antenna in meters.

> = wave length in meters.

AN/MPS-14 RADAR

$$L_{1} = 9.13 \text{ meters}$$

$$L_{2} = 2.28 \text{ meters}$$

$$a = \frac{L_{1}}{\sqrt{2d}} = \frac{9.13}{\sqrt{2X1150.62 \times 0.107}} = \frac{9.13}{15.7}$$

$$a = 0.582$$

$$b = \frac{L_{2}}{\sqrt{2d}} = \frac{2.28}{\sqrt{2 \times 1150.62 \times 0.107}} = \frac{2.28}{15.7}$$

$$b = 0.145$$

2d STEP

From Figure C-2 in Technical Order 31-1-80

The value of F_1 , corresponding to a value 0.582 for "a" is 1

The value of F_2 , corresponding to a value of 0.145 for "b" is 1

$$F_f = F_1 \times F_2$$

 $F_f = 1 \times 1 = 1$

Power Density = 0.178 X 10 $^{-3}$ X 1 = 0.178×10^{-3} Watts/CM²

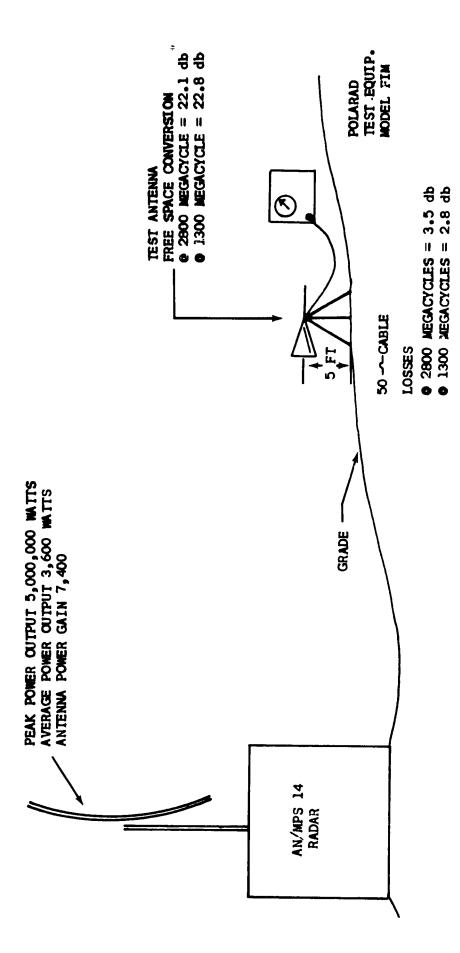


FIGURE 5. Test Instrumentation.



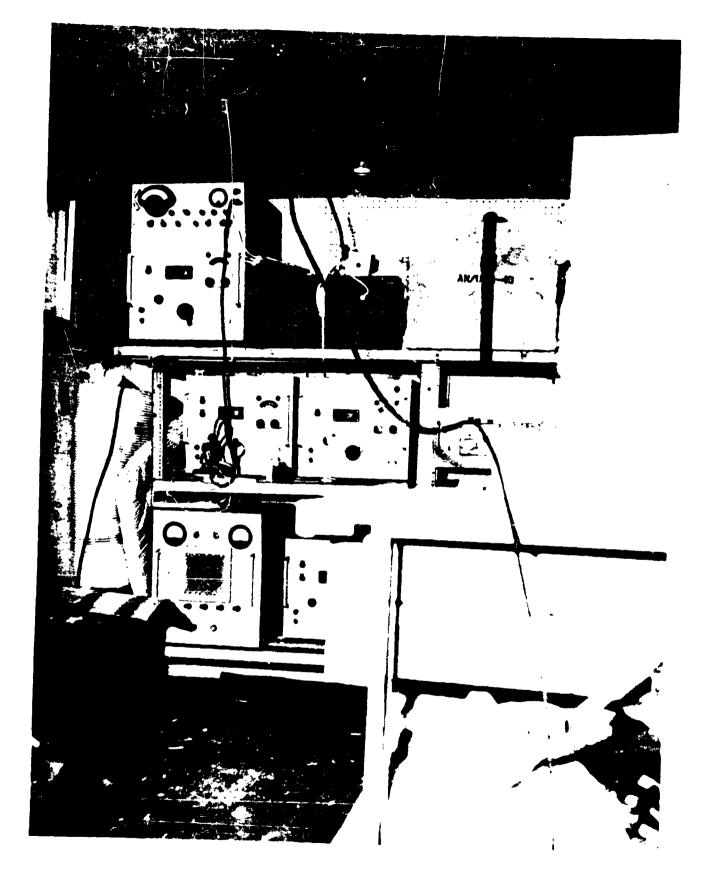


FIGURE 7. Incide Test Van.

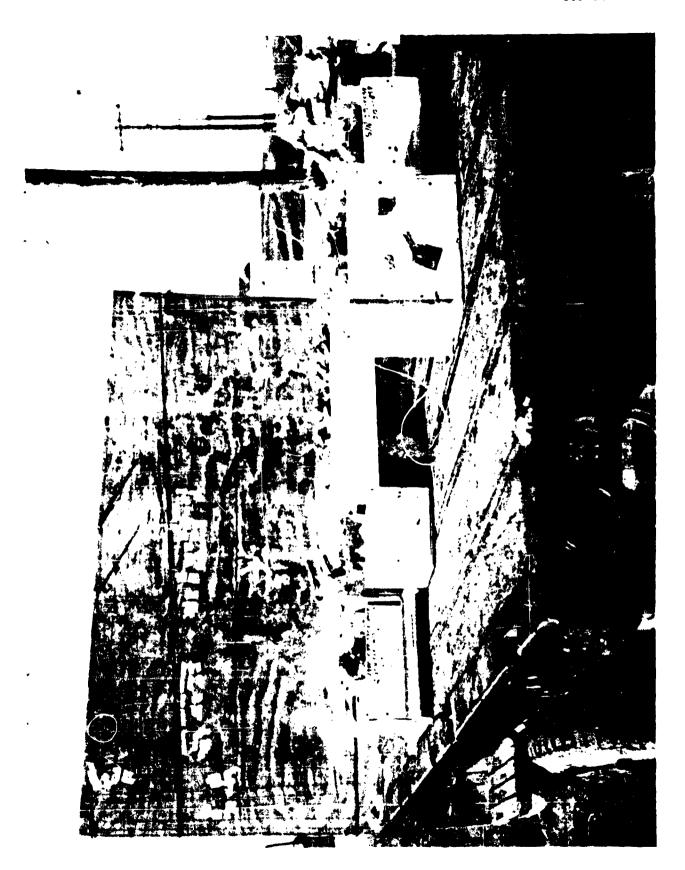
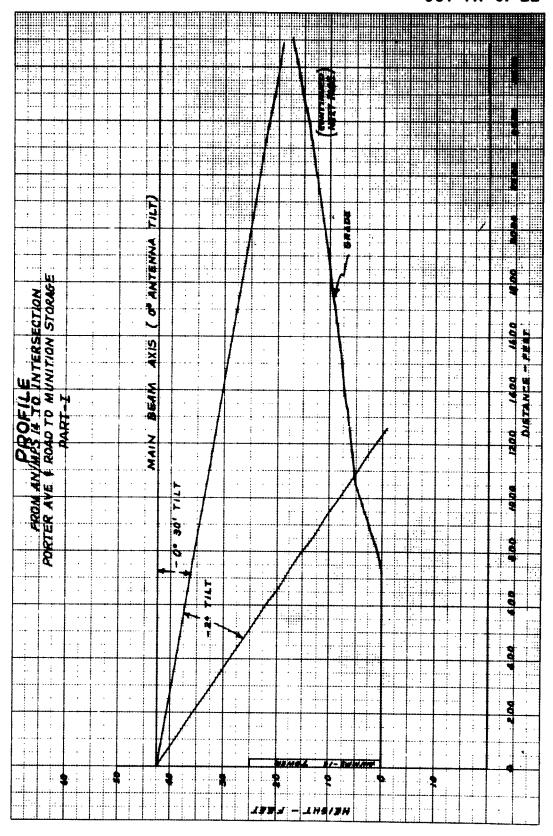




FIGURE 9. Top View of Electrical Initiated Devices as Exposed to R.F. Energy.



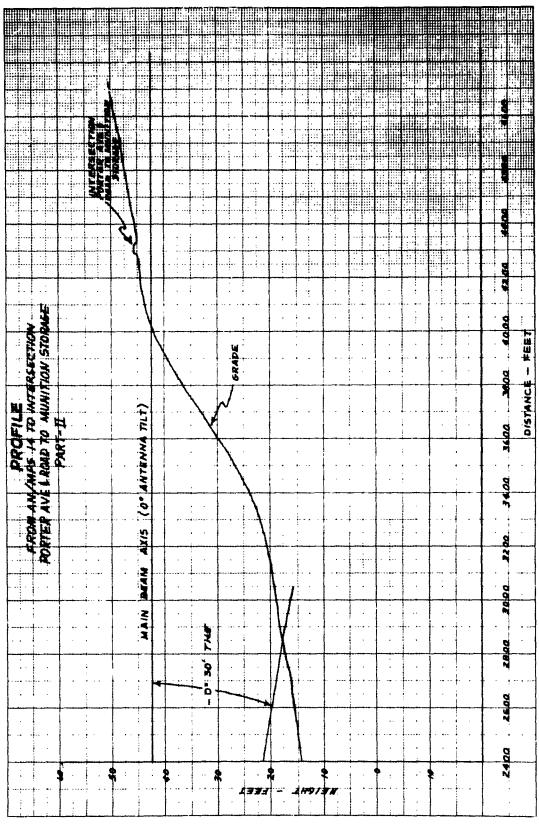


FIGURE 11.

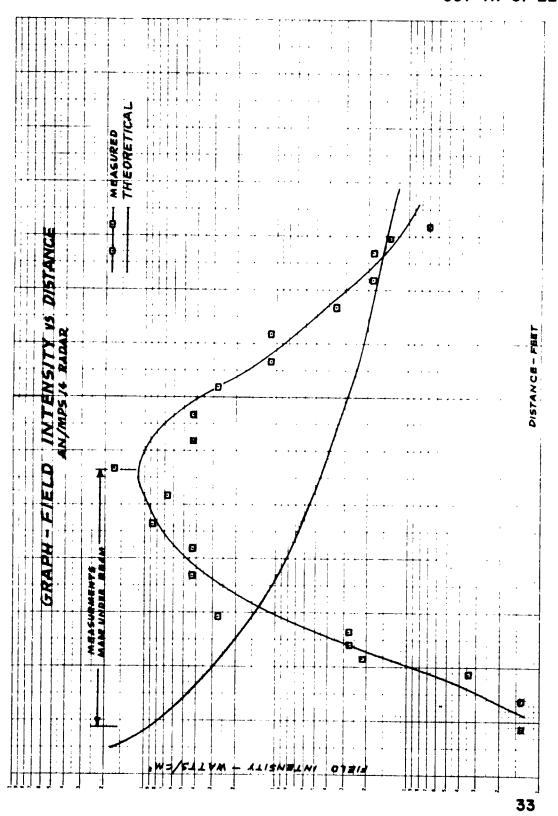


FIGURE 12.

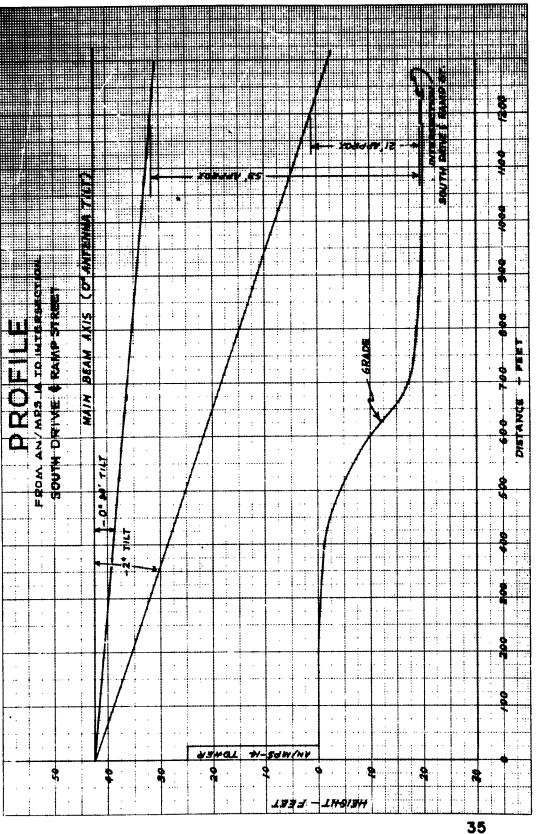


FIGURE 13.

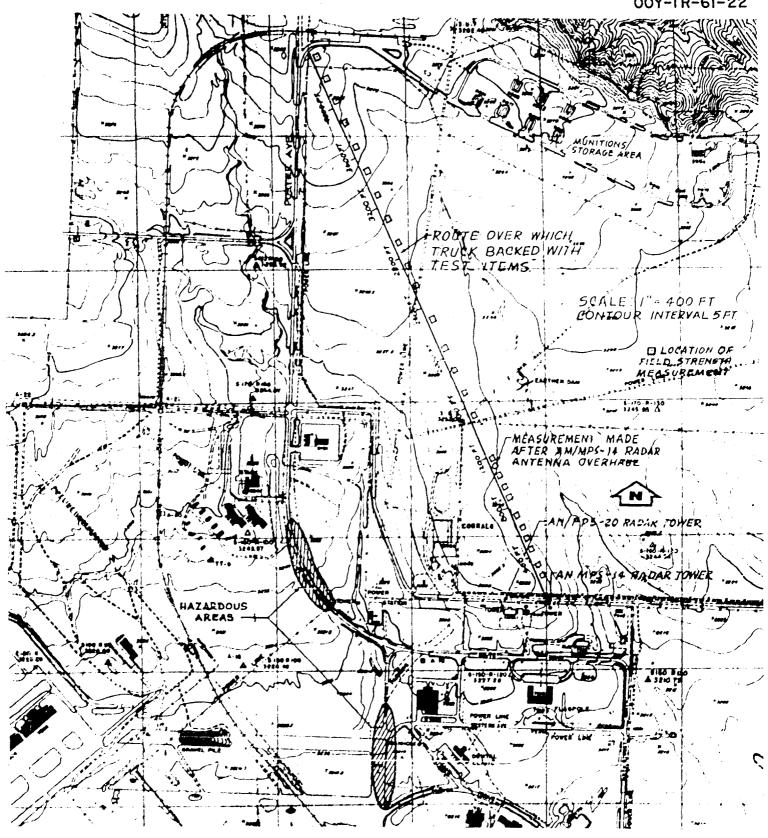


FIGURE 14.

200	200			2 P							THEORE TICAL	r.
OF FIELD STRENGTH	FROM		F. W.	STON		i	į	7	IN TTS/CH ²	PAR.	MEAR FIELD	Callyson
MEASURE-	TOWER (FEET)	RADAR	3 2	7 AC 100	SSOI	dbu/=	PEAK PEAK	PEAK	x 10-3	x 10-3	110	M TTS/22
Front of	3575	AN FPS-20	21	22.8	2.7	169.9	188.4	0.00941	0.0067	0.0934	1	x 10-1
Frant of t	1575	AN/MP9-14	148	22.1	3.5	173.6	478.6	0.0608	0.0438	0.178	1	0.178
190 of 1	3475	AN/ 175-20	177	22.8	2.7	166.5	211.3	0,0118	0.0085	1660*0	1	0.0991
Tont of	3475	AN/MPS-14	3	2.1	3.5	175.6	602.6	0.0963	0.0693	0.189		0.189
Tour of I	3200	AN/FP9-20	133	22.8	2.7	138.5	2 .1	0.00188	0.0014	0.1161	1	
ront of	3200	AN/1899-14	3	2.1	s:	173.6	478.6	9090.0	0.0438	0.223	1	0.223
roat of	3625	AN/FPS-20	140	2.8	2.7	165.5	166.4	0.00941	0.0068	9060*0	1	
Front of	3625	AW/MPS-14	ដ	22.1	3.5	175.6	602.6	0.0963	0.0693	0,.73	1	0.173
1 de 9003	3875	AN/FPS-70	35	2.8	2.7	161.5	104.4	0,00289	0.0021	0.0793	1	
Folia of	3875	AN/1009-14	Ş	22.1	3.5	175.6	602.6	0.0963	0.0693	0.152	1	0.152
Rushmore Gate	4430	AN/FPS-3.	9	2	2.7	165.5	188.4	0.00941	0.0068	0.0607	-1	0.0607
Rushmore Gate	4430	AX/IPS-14	146	22.1	3.5	173.6	478.6	9090*0	0.0408	0.115	1	0.115
Intersection Porter Ave & Road to Rush-	3350	AN/MP9-14	151	z.1	3.5	176.6	676.1	0.1212	0,0873	0.303	-	0.303
nterection	230	AN/FPS-20	133	22.8	2.7	154.5	33.1	0.000748	0.0003	0.762	-	5.762
South Drive &	1250	AN/MPS-14	138	22,1	3.5	153.6	151.4	0,00608	0,0044	1.46	4	1.46
From Inter-		AN/MPS-14	130	2.1	3.5	175.6		0,0963	0.0693	9.138	-	5,138
section of	3971.5	AN/169-14	3	22.1	3.5	178.6		0.192	0.138	0	-	200
Porter Ave	3871.5	AN/MP9-14		23.1	3:5	9.6	955.0	0.242	0.174	200	_	200
F Road to	3671.5	AWARTE 14	7	22	200	3,50	923.0	0.242	0.174	88		0.189
Storage Area	377.5	AN/MONTH	1	22.1	3.5	187.6	28.0	1.527	1.399	0.213	-	0.213
	3071.5	AN MOS-14	162	2.1	3.5	187.6	2399,0	1.527	1.099	0.241	4	0.241
	2871.5	AN/MP9-14	166	22.1	3.5			3.834	2.76	0.2%		0.276
	2671.5	AN/MP9-14	168	22.1	3.5			6.076	4.37	0.319	-	0.315
	2471.5	AW	3	20.1	5	9.8	0.00	24.102	17.4	0.40		0.575
	2071.5	AN/MP9-14	2,1	22.1	3.5		0.9209	9.632	6.93	0.531	-	0.331
	1871.5	AN/MP9-14	171	22.1	3.5	196.6		12.125	6.73	0.631	96.0	0.651
	1671.5	AN/1899-14	168	22.1	3.5	193.6	4786.0	6.076	4.37	0.83 1.83	%.0	0.815
	1471.5	ANVINOS-14	3	22:1				6.076	4.37	g ;	4 6	18.0
-	1071.5	AN/ 89 12	8 2	22.1	3.5	181.6	1202.0	0.3832	0.276	8	8.0	1:331
	971.5	AN/109-14	ட	22.1	3.5	181.6	1202.0	0.3832	0.276	2.42	98.0	2,13
	671.5	AN/MPS-14		22.1	3.5			0.3048	0.219	3.00	0.86	2.58
	771.5	AN/MP9-14	147	22.1	3.5		426.0	0.0481	0.0346	3.8	0.83	3.18
	671.5	AN/16-14	9	Zi.	3.5			880	0.00691	8 9	, c	3.53
	2,100	AN/ 800 14	2 2	32.1		2 3	160 2	0.0192	0.0136	2 5	1	2 2
	371.5	AN/MPS-14	1	22.1	3.5	5.83	269.5	0.0192	0.0136	16.5	S	8.73
	271.5	AM/MP9-14	143	22.1	3.5	166.6	269.5	0.0192	0.0138	30.9	0,40	12.36
:	1130.0	AX/1895-14			3.0db				3.24	1.76	0.925	1.63

*Last Point where field strength metar directly in sxis of reder been. TABLE 1. Electremegnetic Mezard Survey at Elleworth Air Force Base, South Dakota - Data and Calculations.

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ZYOOTH Airmunditions Wing (COMMA), Hill Air Force Base, Utah EMENTHMACE MASS, TO EXPLOSIVES STAVET OF ELISMOHM AIR FORCE MASS, SOUTH DANDIR, by Marcid R. Laughter, June 1861, 44p Incl. fforce TR-51-22) (OOF-TR-51-22) The purpose of this survey was to determine the extent of electromeg- metic redistion hazares to explicative in storage, handling and shipping areas in the vicinity of the A.C. and M. size at Elimorth Air Force Base, South Dantier 7 Ried strength measurements of the main beam from the AM/FPP-O and AM/BP-14 redists were made at various locations in the sforcementioned areas and along the transportation route, Various elec- trically infitted applicative Limas were exposed to the main beam of the states. It was concluded that a degree of hazard does exist along the transportation route to the munitions storage area.	1. Electromegratic Ravisatio I. Harold P. Laughter	2705th Altrenditions Wing (CGAMA), Hill Air Force Base, Ut., EL. STROMORE IIC GALIATION HAZADOS IT EXTLOSIVES SIRVEY OF ELISBORY AIR FOCK BAR, 3001M DAKO.A, by Maroid W. Laughter, June 1961, 41p inci. (Gramma 1961, and the statement of electroments of particular the statement of electroments resist radiation hazaria to sepalestwa in screege, hearling and shipping mests and the vicinity of the A.C. and W. sites at Elizaboth Air Force Base, South Dakots. Field strength measurements of the main beams from the AAN FF-20 and AAN EFS-14 radars were made at various locations in the aforementationed steam and manaportation routs, which we also trivially initiated applicate from the result notes the main beam of the relation to the ANGE-14 radar system. It was concluded that a degree of hazard does exist along the tramportation routs and the relations at the relations to the maintions storage area.	1. Electromagnette Radiation I. Harold F. Laughter
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